# PAPER Circularly Polarized Rounded-Off Triangular Microstrip Line Array Antenna

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**SUMMARY** The Japan Aerospace Exploration Agency (JAXA) plans to launch a geostationary satellite called Engineering Test Satellite VIII (ETS-VIII) in FY 2006. In this paper, a microstrip line array antenna, which has a very simple structure, is introduced to radiate a circularly polarized wave aiming at ETS-VIII applications. This antenna consists of a triangular conducting line with its vertexes rounded off, located above a ground plane, with a gap on one of its side to produce a circular polarization. The proposed antenna is analyzed by numerical simulations for a single element as well as for a three elements array configuration and the possibility of beam-switching in the azimuth space is experimentally confirmed in the latter case. It is found that by properly feeding the elements constituting the array antenna, for an elevation angle  $El = 48^{\circ}$  in Tokyo area, three beams are created in the conical-cut direction with a minimum gain more than 6.6 dBic and an axial ratio less than 3 dB.

key words: microstrip line antenna, circular polarization, triangular shape, rounded vertex, array configuration, beam-switching, mobile satellite communications

# 1. Introduction

The interest in the mobile satellite communications system technologies is expected to increase with the launch of the Japan Aerospace Exploration Agency's (JAXA) Engineering Test Satellite VIII (ETS-VIII). Among its applications, ETS-VIII will conduct orbital experiments on mobile satellite communications in the S-band [1]. One of the experimental aims is the development of a technology enabling the transmission and reception of multimedia information such as voice and images by use of the geostationary satellite for land mobile systems. Although the required specifications depend on the services and systems available, a small and light antenna having a high gain on a wide angular range is desired.

Up to this point, several antennas able to meet these requirements have been extensively investigated, are widely available in the literature [2], and include the conical beam antennas and the satellite-tracking antennas. The attractive feature of the former antenna design is that, as the radiation is omnidirectional in the conical-cut direction and the beam

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is broad in the elevation plane, satellite tracking is not necessary in the elevation plane. However, high gain cannot be achieved because of the isotropy in the conical-cut direction. In contrast, one of the advantages of the latter antenna is that the beam generated by satellite-tracking systems is always directed towards the satellite position even when the azimuth of the mobile station changes [3]. Therefore, such antennas have the possibility to reach a higher gain as compared to the conical beam antennas.

Hence, the authors have researched and presented simple on-board satellite-tracking circularly polarized array antennas for single [4], [5] and dual frequency [6]–[8], aimed at ETS-VIII applications. However, although the aforementioned antennas can satisfactorily be used in outdoor experiments [8], their size is relatively large. Consequently, in this paper, the authors propose a more compact satellite-tracking microstrip line array for ETS-VIII based on the notion of wire antenna [9].

Here, a microstrip line array antenna which has a very simple structure is introduced to radiate a circularly polarized wave. This antenna consists of a triangular conducting microstrip line, whose tips are rounded off, placed above a ground plane and with a gap on one of its side to produce a circular polarization [10]–[14]. The tips are rounded off so that a uniform current distribution is ensured and an impedance mismatch around the vertexes avoided. The proposed element and its array configuration are analyzed by numerical simulations and the possibility of beam-switching in the azimuth space is experimentally confirmed.

# 2. Specifications and Objectives

Table 1 shows the specifications and objectives of an antenna for mobile satellite communications aimed at ETS-VIII applications [4]–[8]. In this research, a thin miniaturized antenna for ETS-VIII and designed for a hundred kbps data transfer is analyzed. In this case, the necessary minimum gain on the azimuth space is set to more than 5 dBic with a maximum axial ratio over the investigated angular range set to less than 3 dB [6]–[8]. In addition, in this study, the measurements are assumed to take place in the center of Tokyo (elevation angle *El* of the geostationary satellite 48°). Furthermore, the operating frequency is fixed at 2.5025 GHz.

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SPECIFICATIONS				
	Transmission	2655.5 MHz		
Frequency	(Tx)	to 2658.0 MHz		
bands	Reception	2500.5 MHz		
	(Rx)	to 2503.0 MHz		
Polarization	Left-handed circular polarization			
	for both Tx and Rx			
OBJECTIVES				
	Elevation angle	48°		
Angular	(El)	(Tokyo)		
ranges	Azimuth angle	0°		
	(Az)	to 360°		
Minimum gain	more than 5 dBic			
Maximum axial ratio	less than 3 dB			

 Table 1
 Specifications and objectives of the antenna for ETS-VIII.

# 3. Structure of the Antenna

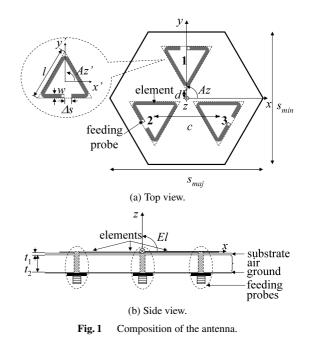
Figure 1 shows the configuration of the triangular microstrip line array antenna, whose tips are rounded off, placed above a ground plane. It is necessary to round off the tips of the triangular microstrip line, else the current launched from the feeding probe will be reflected by the vertexes as their inner angle is sharp and this would generate a standing wave that perturbs the current flow. Consequently, by rounding off the tips of the triangular line, a uniform current distribution is ensured and an impedance mismatch around the vertexes avoided. Although various shapes can be considered [10]–[23], the triangular one is chosen so that it can easily be made and the current distribution on the line can be controlled by changing the gap width and the perimeter. The center of the feeding probe (SMA connector with a long inner part) that connects the ground to the antenna element is located at 1 mm from the gap.

In order to induce circular polarization and sufficient gain, the antenna comprises three single-fed elements with a gap in one of their sides. The elements of the antenna are sequentially rotated [24] by  $120^{\circ}$  so that the gap is located outwards compared to the center of the array ensuring no adverse influence on the adjacent elements that would decrease the performances. Moreover, by sequentially rotating the elements, the relative phase between them is physically shifted. Therefore, assuming the phase of the first one is  $0^{\circ}$ , when rotating the others of  $120^{\circ}$  relatively to the center of the array antenna, phases of elements 2 and 3 are  $120^{\circ}$ and  $240^{\circ}$ , respectively. Such a sequential rotation ensures the generation of circular polarization. Furthermore, by use of the sequential rotation technique, there is a possibility of miniaturization of the antenna.

Figures 1(a) and (b) show the top view and side view of the designed array, respectively. Here, the antenna is made of a thin conducting microstrip line of width w etched on a high relative permittivity substrate whose thickness is  $t_1$ . Additionally, one element has a side l. Hence a perimeter Lgiven by

$$L = 3l - \Delta s \tag{1}$$

where  $\Delta s$  is the gap width. The distance between the center



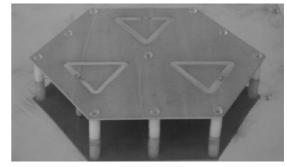


Fig. 2 External view of the fabricated antenna.

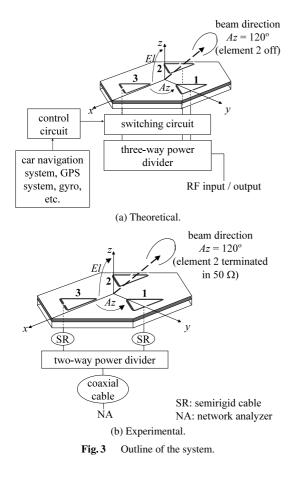
of the array and the tip of a triangle is d. The distance c between two centroids is then

С

$$=\sqrt{3}(d+l/\sqrt{3})$$
 (2)

The ground plane and the substrate are separated by an air layer whose height is  $t_2$ . The height uniformity is ensured by means of plastic spacers. Finally, the substrate size has a hexagonal shape with its major and minor diagonals being  $s_{maj}$  and  $s_{min}$ , respectively.

The antenna is placed for the measurements on a ground plane whose size is 200 mm × 200 mm. An external view of the fabricated antenna is displayed in Fig. 2 and the beam-switching arrangement is shown in Fig. 3. Figure 3(a) shows the theoretical concept and Fig. 3(b) shows the process followed during the measurement. Switching is realized by successively turning off the feed sources of elements 1 to 3. The direction of the created beam is shifted in the azimuth plane by  $-90^{\circ}$  from the element that is turned off for a left-handed circularly polarized antenna [4]–[8]. In Fig. 3, as an example, if element 2 is turned off (experimentally terminated in  $50\Omega$  as explained in the next section), a wide beam is created in the direction  $Az = 120^{\circ}$ . The



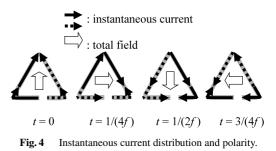
same method is applied to the other elements (1 and 3 alternatively off). As a result, switching all three beams in the conical-cut direction for each element can cover the whole azimuth range.

Information concerning the satellite position can be obtained by use of systems such as car navigation, GPS, gyro, etc. Although not discussed in this paper, the electronic circuit that controls the antenna and selects the elements to be turned on or off, makes use of the previously obtained information. In order to control the elements, switching circuits based on PIN diodes [25], [26] or switching ICs can be used.

## 4. Performance of the Antenna

The Method of Moments (MoM) has been chosen in the analysis for its asset of fast calculation. Owing to the software characteristics, the dielectric substrate and the ground plane are considered to be infinite. As the microstrip line antenna is backed by a perfectly conducting ground plane of an infinite extent that acts as a reflector, this antenna can be handled by the method of image.

In the next subsections, the condition for circular polarization is explained at first. Then a single element and the case of the three elements array antennas are analyzed and experimental results are shown in the case of the array. Precisely, the frequency characteristics of the impedance and antenna performances are shown. Finally, the results in



terms of gain and axial ratio in both the elevation plane and the conical-cut direction are presented. The performances are compared with measurements realized in the radio anechoic chamber of the Graduate School of Science and Technology, Chiba University, Japan.

It should be noted that during measurements, the actual switching network was not included because this paper focuses on the fundamental behavior and characteristics of the antenna. Instead, the switching process was performed by manually changing the elements to be fed or not, as explained in Sect. 3. Practically, when element 2 was investigated, elements 1 and 3 were connected to a two-way power divider by means of semi-rigid cables and element 2 was terminated in 50  $\Omega$  (see Fig. 3(b)).

# 4.1 Condition for Circular Polarization

Before dealing with the array configuration, it is necessary to have a traveling wave current distribution which has a constant amplitude and a linearly changing phase for a single loop antenna to radiate a circularly polarized wave. In the case of a circular loop, a circularly polarized wave can be radiated by means of loading a reactance of an appropriate value [15]. Instead of loading an appropriate reactance, a very simple method is to use a gap [10]–[14]. It has been shown theoretically that by introducing such a gap with a certain width on the printed element and feeding the antenna with a coaxial probe, a traveling-wave current distribution could be excited, and as a result, a circular polarization can be achieved.

Fundamentally, using the line electric-current source method, the way of the proposed antenna to radiate circular polarization can be easily understood. Assuming one element to be fed from the bottom left part compared to the gap location as in Fig. 1. The instantaneous current distribution is shown by the arrows in Fig. 4 where the arrows point in the opposite direction every half guide wavelength  $\lambda_a$ . From this figure, at time t = 0, the total radiation field from each segment consists of the juxtaposition of the current flowing on the left and right segments of the triangular line, because the radiated fields of the horizontal segment at the bottom cancel each other out. Hence the radiated field is upwards directed in the plane where the element lays. When t = 1/(4f), where f is the frequency, the total radiation field will be of horizontal right-oriented polarity. Similarly, when t = 1/(2f) or t = 3/(4f), the polarity of the total radiation field will be oriented downwards or to the left, respectively. As shown in the figure, the polarity of the total field of the radiated electromagnetic waves (normal to the surface of the paper), rotates clockwise, and completes one cycle in time 1/f. Thus, the fundamental element of the antenna operates as a left-handed circularly polarized antenna.

# 4.2 Application to a Single Element

# 4.2.1 Composition

The theory described above is confirmed after by numerically analyzing a single element thus proving the generation of a circularly polarized wave. The parameters, deduced from various analyses not explicitly shown here for reason of shortening, are summarized in Table 2. In order to show the influence and necessity of rounding off the tip of the triangular microstrip line, two cases are numerically analyzed: triangular microstrip with and without rounded vertexes.

# 4.2.2 Input Characteristics

Figure 5 shows the S-parameters. From this figure, the Sparameters tend to improve when the vertexes are rounded off. It is due to the fact the impedance mismatch is avoided or at least decreased around the rounded tips, as previously stated. Indeed by rounding the tip, the line width w is always constant even around the vertex, when the value of the line width would be 2w in case the vertex is straight.

 Table 2
 Antenna parameters (single element).

	Symbol	Value (straight vertexes)	Value (rounded vertexes)
Relative permittivity	$\epsilon_r$	9.80	9.80
Loss tangent	$tan \delta$	0.003	0.003
Substrate thickness	$t_1$	1.27 mm	1.27 mm
Air thickness	$t_2$	14.00 mm	14.00 mm
Element side	l	39.14 mm	40.54 mm
Line width	w	3.00 mm	3.00 mm
Gap width	$\Delta s$	1.00 mm	1.00 mm

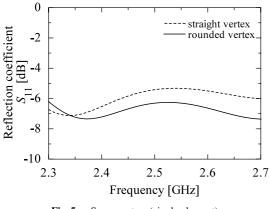


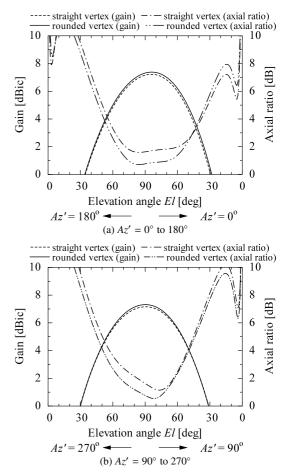
Fig. 5 S-parameters (single element).

#### 4.2.3 Radiation Characteristics in the Elevation Plane

Figure 6 represents the radiation characteristics in the elevation plane for a single element in terms of gain and axial ratio. Figure 6(a) shows the results in the  $Az' = 0^{\circ}$  to  $180^{\circ}$  plane while Fig. 6(b) shows the performances in the  $Az' = 90^{\circ}$  to  $270^{\circ}$  plane. From both figures, it is obvious that the use of rounded vertexes improves the antenna performances. In Fig. 6(a), the minimum axial ratios are 1.6 dB and 0.7 dB for straight and rounded vertexes, respectively. An improvement of 5° in the 3 dB-axial ratio beamwidth can be observed as well. In Fig. 6(b), the minimum axial ratio drops from 1.1 dB to 0.5 dB with an increase of 10° in the 3 dB-axial ratio beamwidth when the vertexes are rounded off. From these results, it can be said that the use of rounded vertexes improves the current distribution.

As a conclusion, it has been shown that a single triangular microstrip line antenna element radiates a circularly polarized wave when its constituting tips are rounded off, ensuring a uniform current distribution and an impedance mismatch avoidance.

In the next step, the single element antenna is put into an array configuration and the obtained results are presented in the following section.





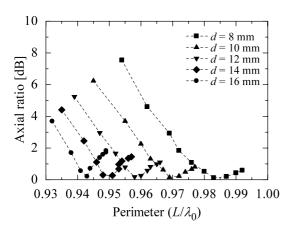
#### 4.3 Case of the Array Antenna

#### 4.3.1 Parameters Analysis

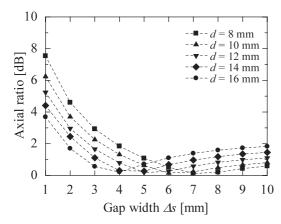
It has been shown above that a single element can radiate a circularly polarized wave provided that proper parameters are used. In this section, the case of the array is analyzed in details and it is clarified here that suitable parameters for the perimeter of the line, the gap width, the distance between adjacent elements, and the air thickness exist to radiate a circularly polarized wave.

The parameters of the line considered are deduced from the variations of the perimeter *L*, the gap width  $\Delta s$  and the distance between the center of the array and the tip of one element *d*. Figures 7 and 8 show the axial ratio characteristics as a function of *L* and  $\Delta s$ , respectively. In this research, the air thickness  $t_2$  is fixed to 14 mm (0.12 $\lambda_0$ ) as for the case of a single element in order to decrease the number of parameters to be analyzed. In these graphs, the axial ratio is examined in the case  $El = 48^{\circ}$  and  $Az = 120^{\circ}$  (i.e. element 2 off).

Figure 7 shows the axial ratio when L is changed for various values of d. In the present case, the value of  $\Delta s$ 



**Fig.7** Axial ratio characteristics vs. perimeter ( $Az = 120^\circ$ ,  $El = 48^\circ$ ).



**Fig. 8** Axial ratio characteristics vs. gap width ( $Az = 120^\circ$ ,  $El = 48^\circ$ ).

is fixed to 3.0 mm. From this figure, good axial ratios for circular polarization are obtained for appropriate values of L. In addition, it can be seen that the perimeter decreases with an increase in d. This is explained by the fact that the mutual coupling between elements decreases along with the increase in d.

Figure 8 shows the variation of axial ratio when  $\Delta s$  is changed. Here, the value of *L* is optimized every time so that the axial ratio is minimal at the targeted operating frequency and the so obtained value of *L* is used for each  $\Delta s$ . From this figure, it can be seen that the axial ratio decreases until a minimum value then increases with an increase in  $\Delta s$ . It comes from the fact the antenna gets very inductive with an increase in  $\Delta s$ , which induces a decrease in the axial ratio as stated in [15]. Hence the value of  $\Delta s$  should be chosen neither too large nor too small.

From these graphs, aiming at an element and antenna global size as small as possible (i.e. the antenna should be contained within a diameter of 100 mm from the array center to the bottom edge of an element), as well as an input impedance where  $X_{in}$  is as close as possible to 0 and axial ratio performances as low as possible (typically less than 2 dB), the array antenna whose parameters are summarized in Table 3 is numerically and experimentally analyzed in the next sections. Use of the proposed configuration allows a decrease of about 20% in the surface of a single element compared to previously [6]–[8], with the distance *c* between two centroids being about  $0.51\lambda_0$ , where  $\lambda_0$  is the wavelength in free space.

# 4.3.2 Input Characteristics

Figures 9 and 10 show the input characteristics of the antenna, i.e. S-parameters and input impedance, respectively. The input characteristics of the antenna are measured at the input of the probe that is soldered to the ground (i.e. the admittance of the probe is not included in the result) in case of element 2 only, or at the input of the power divider in the case of the array (element 2 terminated in 50  $\Omega$ ). In Fig. 9, the measurement and the simulation present the same tendency. The measurement results are slightly better than the

Table 3	Antenna parameters	(array configuration)	

Table 5 Antenna parameters (array configuration).				
	Symbol	Value		
Relative permittivity	$\epsilon_r$	9.80		
Loss tangent	$tan \delta$	0.003		
Substrate thickness	$t_1$	1.27 mm		
Air thickness	$t_2$	14.00 mm		
Element side	l	40.76 mm		
Line width	w	3.00 mm		
Gap width	$\Delta s$	3.00 mm		
Distance between center of the array and tip of the element	d	12.00 mm		
Substrate major diagonal	s <sub>maj</sub>	140.00 mm		
Substrate minor diagonal	S <sub>min</sub>	120.00 mm		

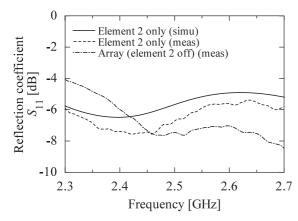


Fig. 9 S-parameters (array configuration).

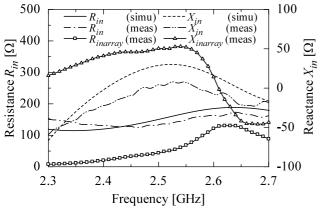
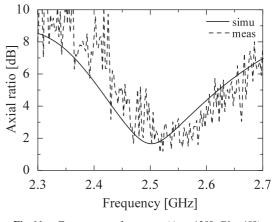


Fig. 10 Input impedance (array configuration).

simulation due to the accuracy in the fabrication of the antenna. Although not explicitly shown here, the input characteristics tend to be better with a decrease in  $t_2$  to a certain extent. During the fabrication, it appears that  $t_2$  was slightly lower than its simulated value hence explaining the improvement. In addition, from the same figure, when the power divider is used, improvement is realized due to the inner properties of the power divider, which is a Wilkinson divider.

In Fig. 10, the resistance and reactance are shown. The full line and dashed line represent the real part (resistance  $R_{in}(\text{simu})$ ) and the imaginary part (reactance  $X_{in}(\text{simu})$ ) by simulation while the dash single-dotted line and the dash double-dotted line show their counterpart  $R_{in}$  (meas) and  $X_{in}$ (meas) obtained by measurement. Additionally, in the case of the array (element 2 off), the square and triangle plotted lines represent the resistance  $R_{inarray}$  (meas) and reactance  $X_{inarray}$  (meas), respectively. Only the case of element 2 is presented for reason of graph legibility. The slight improvement in the results of measurement compared to the simulation can be explained by the same reason as above. From this figure, it appears the input impedance is about  $150\,\Omega$  in the case of an element measured independently. During the measurement of the radiation patterns, as the inactivated element is terminated by 50  $\Omega$ , a mismatch can



**Fig. 11** Frequency performances ( $Az = 120^\circ$ ,  $El = 48^\circ$ ).

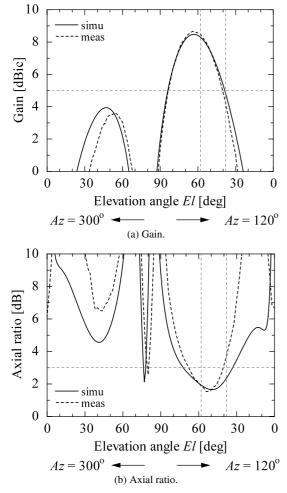
occur and the phenomenon of reradiation from the terminated element that acts as a parasitic might happen. However, for actual use, the antenna will eventually be backed by a switching circuit to control the element to be turned off. Hence even if the antenna impedance is not  $50 \Omega$ , a method to eliminate the problem of reradiation is to use  $150 \Omega$  lines for the switching circuit rather than  $50 \Omega$  lines.

# 4.3.3 Frequency Characteristics

The antenna was numerically optimized by minimizing the axial ratio at  $El = 48^{\circ}$  based on simulations for the targeted receive frequency as shown in Fig. 11, in the case element 2 off. The axial ratio is 1.7 dB at 2.5025 GHz. In the same figure, it can be seen that measured results follow the same tendency with a shift of 15 MHz in the upper frequency and a narrowing in the bandwidth. This shifting is explained by the influence of the ground plane which is finite in the case of measurement while it is infinite by numerical simulations [4], [5]. Note that the rippling can be explained by the ground plane effect and by the fact the number of points used for the averaging might have been too small.

# 4.3.4 Radiation Characteristics in the Elevation Plane

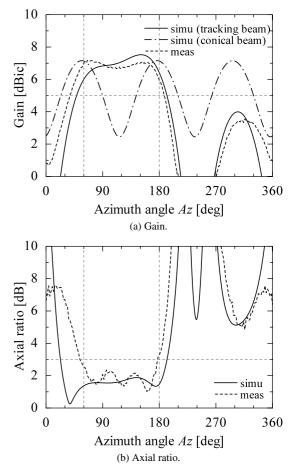
Figure 12 shows the radiation characteristics of the antenna in the elevation plane when element 2 is off (in the direction  $Az = 120^{\circ}$  to 300°). The full line and dashed line represent the results obtained by numerical simulation and measurement, respectively. Figure 12(a) shows that the maximum of radiation occurs when  $El = 63^{\circ}$  for the numerical simulation and  $El = 64^{\circ}$  for the measurement. Additionally, in Fig. 12(b) it can be observed that the axial ratio obtained by numerical analysis and by measurement is satisfactorily less than 3 dB at 48°. The beam narrowing, especially in the case of the axial ratio, is due to the influence of the finiteness of the ground plane.



**Fig. 12** Radiation characteristics in the elevation plane ( $Az = 120^{\circ}$  to  $300^{\circ}$ ).

# 4.3.5 Radiation Characteristics in the Conical-Cut Direction

Figures 13(a) and (b) represents the gain and axial ratio characteristics of the antenna for  $El = 48^{\circ}$  in the case element 2 turned off. As previously, the full line and dashed line represent the results obtained by numerical simulation and measurement, respectively. In addition, the performances obtained by simulation in the case of a conical beam are plotted in dash dotted line for comparison. Figure 13(a) shows that by numerical results, the beam for which the gain is more than 5 dBic has a width of 143°. Additionally, in the same figure, it is seen that the 6 dBic gain beamwidth has a width of 124°. In the case of measurement, it is 148° and  $130^{\circ}$ , respectively. It can be seen here that when all the patches are activated, the gain of the conical beam fluctuates between 2.4 dBic and 7.2 dBic over the whole azimuth space. Moreover, from this figure, it is clear the beamwidth is insufficient to scan the whole azimuth space as requested in the specifications (Table 1) at low or medium elevation angle. Thus, it can be confirmed the use of satellite-tracking

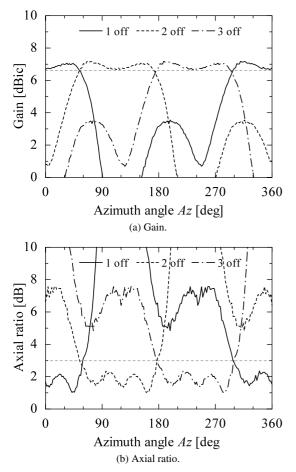


**Fig. 13** Radiation characteristics in the conical-cut direction  $(El = 48^{\circ})$ .

beam is more suitable to the present application as stated in the Introduction. In addition, from Fig. 13(b), the broadness of the axial ratio characteristics was confirmed by simulation and measurement as it is less than 3 dB over a range of 169° and 122° centered on 120°, respectively.

4.3.6 Verification of Beam-Switching in the Azimuth Space

The antenna characteristics were measured by turning off one of the elements from 1 to 3 in turn. By doing so, the beam-switching operation could be observed and confirmed (Fig. 14). Figure 14(a) shows the gain of the antenna in the conical-cut direction when the geostationary satellite is located at  $El = 48^{\circ}$  from Tokyo. Results show that each generated beam is centered at  $Az = 114^\circ$ ,  $234^\circ$ , and  $354^\circ$  in the conical-cut direction with respect to the element that is turned off. Thus, the existence of three beams with 120° difference between them in the conical-cut direction is confirmed. In the same figure, it is seen that the minimum gain corresponding to the intersection between two adjacent beams ( $Az = 54^\circ$ , 174°, and 294°) is 6.6 dBic. Additionally, in Fig. 14(b), the maximum axial ratio (corresponding to the maximum value obtained at each intersection of the gain as previously indicated) is 2.9 dB. Therefore, it meets



**Fig. 14** Beam-switching in the conical-cut direction ( $El = 48^{\circ}$ ).

the specification of less than 3 dB over the whole azimuth range.

Previously investigated array antennas [6]-[8], etched on a high relative permittivity substrate, cannot meet the requirements for mobile satellite communications as their surface, hence their gain, decreases too much. However, from the previously discussed results, it was confirmed that with a triangular microstrip line antenna, whose surface from the array center to the bottom edge of an element is 32% reduced compared to the previously investigated array antennas [6]–[8] (corresponding to a reduction of 20% in the case of a single element), equivalent performances could be obtained in terms of axial ratio coverage, minimum gain, etc. Note that the size reduction of the proposed triangular microstrip line antenna is due to the use of a dielectric substrate whose relative permittivity is high and due to the fact that microstrip lines are intrinsically smaller than patch antennas.

## 5. Conclusion

In this paper, both a single element and an array of three elements antennas, composed by a simple triangular microstrip line with a gap, able to generate a circular polarization, have been studied. A circular polarization can be simply obtained by properly adjusting the element parameters. The analysis of the array has shown that the generated beam can be switched in the azimuth space with a minimum gain of 6.6 dBic and a maximum axial ratio of 2.9 dB.

In the future, the input characteristics of the antenna will be improved by use of a microstrip line feeding or parasitic elements to tune the impedance. In addition, improvement of the antenna performances at low elevation will be researched. Furthermore, a dual-band configuration will be investigated.

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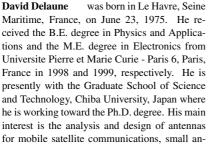
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